













of Shell and Tube Thermal Energy

Exergy, Energy and S Environment Symposium



11th International

Exergy Energy

Environment **Symposium**

July 14-18, 2019

Chennai, INDIA

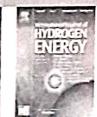
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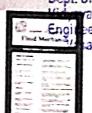
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ISBN: 978-83-5383-279-2

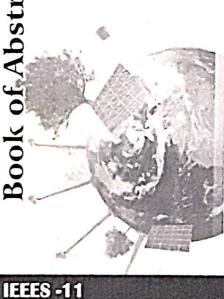
Environment Symposium Exergy, Energy and 11th International **IEEES -11**

and Environment Symposium II" International Exergy, Energy

July 14" - 18" 2019

SRM Institute of Science and Technology, INDIA

Book of Abstracts



ISBN: 978-93-5382-239-2



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Thursday, 18 July, 2019



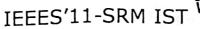
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276: Thermal Performance Analysis of Shell and Tube Thermal Energy Storage UnitUsing Evacuated Tubes Solar Collector

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Abstract:

The paper discussed an experimental work aiming to understand the role of phase change material (PCM) in a thermal insulated shell and tube thermal energy storage unit. The evacuated tubes solar collector (ETSC) used for the experimental setup. The shell and tube thermal energy storage unit consists of a single copper tube with longitudinal amount of working fluid. The heat pipe of ETSC was inserted longitudinally above the single copper tube in shell and tube thermal energy storage unit. During a day period, the intensity of solar radiation falls on evacuated tubes and the PCM starts melting as the heat pipe temperature increases continuously. At the same time, the cold water enters water and PCM observed and an outlet of a single copper tube the temperature of water increases. The experimental readings are carried out in a day period during availability of solar radiation at a different mass flow rate of cold water performance thermal energy storage unit. Finally, collector efficiency and system efficiency obtained in efficiency Vs time plot.

Keywords: Solar Evacuated Tube Collector, Melting, PCM, Shell and Tube Thermal Energy Storage Unit.

277: COMPUTATIONAL STUDY OF SAFE SEPARATION OF SABOT FROM PENETRATOR IN APFSDS

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Abstract:

The paper addresses the computational fluid dynamics solution for safe separation of sabot from the penetrator in armour piercing fin stabilised discarding sabot. The design and functional characteristics of sabot are modeled to achieve a safe trajectory. Driving bands are important components of APFSDS which adds the sabot together to efficiently transfer the gas forces on penetrator. Sabot, penetrator, fins and driving bands are modelled separately which further assembled together using solid works CAD software. All tri element is used to generate the un-structured surface mesh of the model using ICEM software. The aerodynamic coefficients of lift, drag and moment acting on the penetrator and sabot discard process are evaluated by using in house developed finite volume based implicit solver. The discarding process of sabot generates shocks at supersonic speed which can affect the trajectory of the penetrator during flight. The timely breaking of bands, after muzzle exit, the APFSDS is an important phenomenon to simulate and govern safe separation of sabots. Based on finite element method software, the breaking of bands was simulated to find the stress and deformation under pressure force exerted on sabot cups. The pressure exerted on sabot cups and separation of sabot due to aerodynamic forces were simulated using in house developed finite volume based implicit CFD solver. The CI, Cd and pitch moment are plotted against the angle of sabot with penetrator which clearly indicates the uniform separation of the sabot.

Keywords: CAD, APFSDS, CI, Cd, CFD, Lift, Drag, Moment, Muzzle, Sabot, Penetrator

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